

# Gas-Turbine Combustion using Hydrogen Enriched Fuels

- ✎ Gas turbines are one of the world's most important sources of industrial and domestic power, they are also the driving force behind aviation.
- ✎ Currently hydrocarbon fuels dominate since they are safe and well understood.
- ✎ Alternative fuels are an increasing issue and Hydrogen is one of the most important.
- ✎ Hydrogen offers large reductions in CO<sub>2</sub> and the elimination of CO and UHCs. Low NO<sub>x</sub> technologies are applicable to H<sub>2</sub> to minimise NO<sub>x</sub> emissions.

# Hydrogen Enriched Fuels - a First Logical Step

- ✎ Complete re-design of current machines may not be desirable and, in any case, may delay the adoption of H<sub>2</sub> fuels.
- ✎ Pure H<sub>2</sub> presents several problems for flame stability due to its very high flame speed when premixed, and its high temperatures when non-premixed.
- ✎ Tailoring a fuel of hydrogen-hydrocarbon allows for a more progressive transition of fuels while still affecting a reduction in carbon dioxide emissions.
- ✎ Tailored fuels may assist the operation of combustors near their lean limit and encourage the use of biomass fuels.

# Technical Challenges for Hydrogen Enriched Fuels

- ✎ The behaviour of multi-component fuels is a complex issue for both the mixing and the combustion cycles.
- ✎ Whilst the behaviour of both hydrogen and hydrocarbon flames is relatively well understood, from fundamental to applied configurations, multi-component fuel mixtures require a great deal of fundamental investigation before being applied.
- ✎ This proposal is an effective way to gain this knowledge through a joint program of kinetics, modelling and supporting experimental work.

## Technical Programme Summary

- ✍ This programme will demonstrate an effective way to gain the essential basic knowledge through a joint program of kinetics, modelling and supporting experimental work.
- ✍ The program will develop a fundamental understanding of hydrogen mixtures, progressively apply it to more applied geometries and finally apply it in realistic gas turbine systems.
- ✍ The activity will comprise of two stand-alone programmes (1 EU and 1 US) which can interlock to maximise mutual benefit.

## Programme Partners & Roles

- ✍ **The EC project would have 3 main elements:**
- ✍ Kinetics managed by *Warnatz*, Heidelberg,
- ✍ Modelling managed by *Janicka*, Darmstadt
- ✍ Diagnostic Experiments managed by *Hutchinson*, & *Greenhalgh*, Cranfield.
  
- ✍ **The US project:**
- ✍ Complementary activities on all three fields above managed by *Gallagher & Keller*, CRF, Sandia National Labs., Livermore US.

## Key Technical Aspects

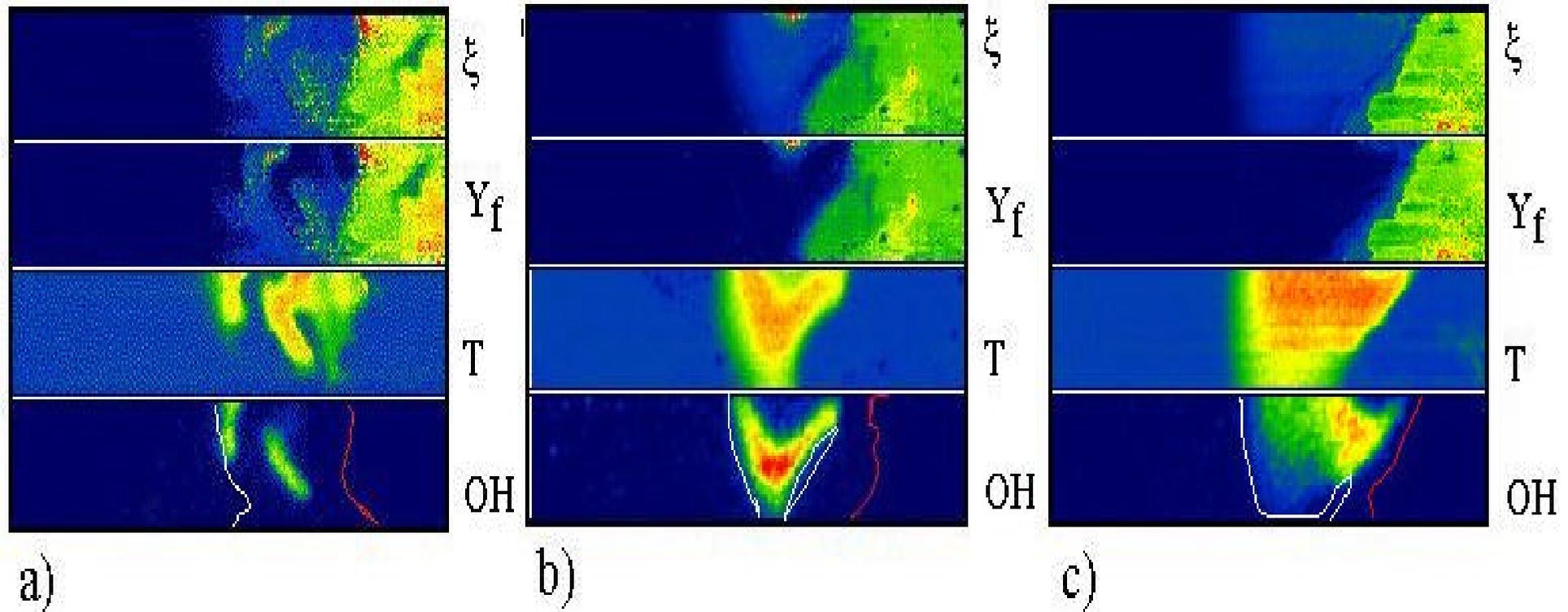
- ✍ **Kinetics** (building on experience from past EC projects):
  - ✍ *Automatic reaction generation (LISP) and mechanism reduction, ignition & extinction modelling, coupling to flow through pdf modelling.*
- ✍ **Modelling:**
  - ✍ *LES modelling with specific SGS combustion sub-grid model development.*
- ✍ **Experiments:**
  - ✍ *Validation experiments for SGS model development and model validation (e.g. scalar dissipation and differential diffusion effects).*
  - ✍ *Progressive validation data from more realistic geometries.*

## Advantages of Combustion-LES

- ✎ The sub-grid-scale (SGS) model improves with finer grids. Thus, a good balance can be achieved between computational effort and accuracy of results
- ✎ LES has the potential to correctly describe unsteady phenomena. This becomes important for gas-turbine combustors where thermo-acoustic instabilities need to be resolved.
- ✎ The classical problem of LES (description of wall effects) becomes irrelevant here since in Combustion-LES, all important aspects are bound to regions off the walls (reaction, heat release, pollutant formation, radiative heat loss).
- ✎ Although combustion happens within the sub-grid and still has to be modelled, it can be described much more accurately because sub-grid fluctuations are small (30% of those in RANS) and unsteady information is available.

# Simultaneous Imaging of fuel, T & OH

Kelmann, Cranfield



3 instantaneous series of images (a-c) of mixture fraction ( $\xi$ , black-red, 0-0.3), fuel mass fraction (black-red, 0-0.3), temperature (black-red, 0-2200K) and OH concentration (black-red, 0-6.9E+16 molecules/cc) in a lifted turbulent methane diffusion flame. Lean reactive limit: ( $\xi = 0.02$ ) white, rich reactive limit: ( $\xi = 0.12$ ), red